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# **SECTION 2**

# LIMITATIONS

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#### **SECTION 2**

#### **LIMITATIONS**

### 2.1 GENERAL

This section provides the "FAA Approved" operating limitations, instrument markings, color coding and basic placards necessary for the safe operation of the airplane and its systems.

Limitations associated with those optional systems and equipment which require handbook supplements can be found in Section 9 (Supplements).

#### 2.3 AIRSPEED LIMITATIONS

SPEED	KIAS	KCAS
Never Exceed Speed (VNE) - Do not exceed this speed in any operation.	190	186
Maximum Structural Cruising Speed (VNO) - Do not exceed this speed except in smooth air and then only with caution.	149	148
Design Maneuvering Speed (VA) - Do not make full or abrupt control movements above this speed.		
At 2750 lbs. G.W. At 1863 lbs. G.W.	121 96	121 97

## **CAUTION**

Maneuvering speed decreases at lighter weight as the effects of aerodynamic forces become  $2.8/10^{\circ}$ more pronounced. Linear interpolation may be used for intermediate gross weights. Maneuvering speed should not be exceeded while operating in rough air.

## **SECTION 2** LIMITATIONS

## PIPER AIRCRAFT CORPORATION PA-28RT-201, ARROW IV

SPEED	KIAS	KCAS
Maximum Flaps Extended Speed (VFE) - Do not exceed this speed with the flaps extended.	108	104
Maximum Landing Gear Extension Speed - Do not exceed this speed when extending the landing gear.	130	130
Maximum Landing Gear Retraction Speed - Do not exceed this speed when retracting the landing gear.	109	109
Maximum Landing Gear Extended Speed (VLE) - Do not exceed this speed with the landing gear extended.	130	130
2.5 AIRSPEED INDICATOR MARKINGS		
MARKING	IA	S
Red Radial Line (Never Exceed)		190 KTS
Yellow Arc (Caution Range - Smooth Air Only)	14	9 KTS to 190 KTS
Green Arc (Normal Operating Range)	5	8 KTS to

White Arc (Flap Down)

58 KTS to 149 KTS

53 KTS to 108 KTS

## SECTION 2 LIMITATIONS

## 2.7 POWER PLANT LIMITATIONS

(b)	Number of Engines Engine Manufacturer	l Lycoming
(c)	Engine Model No.	IO-360-C1C6
(d)	Engine Operating Limits	
	(1) Maximum Horsepower	200
	(2) Maximum Rotation Speed (RPM)	2700
	(3) Maximum Oil Temperature	245° F
(e)		
	Minimum (red line)	25 PSI
	Maximum (red line)	100 PSI
(f)	Fuel Pressure	
	Minimum (red line)	14 PSI
	Maximum (red line)	45 PSI
(g)		100/130 - Green
	Number of Propellers	1
(i)		McCauley or Hartzell
(j)	Propeller Hub and Blade Model	
	(1) McCauley	B2D34C213/90DHA-16
	(2) Hartzell	HC-C2YK-1()F/
		F7666A-2R
(k)	Propeller Diameter	
	(1) McCauley	
	Minimum	73
	Maximum	74
	(2) Hartzell	
	Minimum	72
	Maximum	74
(1)	Blade Angle Limits	
	(1) McCauley	12.5 . 0.20
	Low Pitch Stop	$12.5 \pm 0.2^{\circ}$
	High Pitch Stop	$27.5 \pm 0.5^{\circ}$
	(2) Hartzell	
	Low Pitch Stop	$14.0 \pm 0.2^{\circ}$
	High Pitch Stop	29.0 ± 2.0°

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### SECTION 2 LIMITATIONS

# PIPER AIRCRAFT CORPORATION PA-28RT-201, ARROW IV

(m) RPM Restrictions (McCauley Propeller Only)

Avoid Continuous Operation Between 1500 and 1950 RPM Below 15 Inches Map.

# 2.9 POWER PLANT INSTRUMENT MARKINGS

(a)	Tachometer Green Arc (Normal Operating Range) Red Line (Maximum Continuous	500 to 2700 RPM
	Power)	2700 RPM
(b)	Oil Temperature	
	Green Arc (Normal Operating Range)	75° to 245° F
	Red Line (Maximum)	245° F
(c)	Oil Pressure	
	Green Arc (Normal Operating Range)	60 PSI to 90 PSI
	Yellow Arc (Caution Range) (Idle)	25 PSI to 60 PSI
	Red Line (Minimum)	25 PSI
	Red Line (Maximum)	100 PSI
(d)	Fuel Pressure	
	Green Arc (Normal Operating Range)	14 PSI to 45 PSI
	Red Line (Minimum)	14 PSI
	Red Line (Maximum)	45 PSI

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SECTION 2 LIMITATIONS

#### 2.11 WEIGHT LIMITS

(a) Maximum Weight

2750 LBS.

(b) Maximum Baggage

200 LBS.

#### NOTE

Refer to Section 5 (Performance) for maximum weight as limited by performance.

### 2.13 CENTER OF GRAVITY LIMITS

Weight Pounds	Forward Limit Inches Aft of Datum	Rearward Limit Inches Aft of Datum
2750	90.0	93.0
2400	85.5	93.0

### **NOTES**

Straight line variation between points given.

The datum used is 78.4 inches ahead of the wing leading edge at the intersection of the straight and tapered section.

It is the responsibility of the airplane owner and the pilot to insure that the airplane is properly loaded. See Section 6 (Weight and Balance) for proper loading instructions.

# 2.15 MANEUVER LIMITS

No acrobatic maneuvers including spins approved.

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### SECTION 2 LIMITATIONS

### PIPER AIRCRAFT CORPORATION PA-28RT-201, ARROW IV

#### 2.17 FLIGHT LOAD FACTORS

(a) Positive Load Factor (Maximum)

3.8 G

(b) Negative Load Factor (Maximum)

No inverted maneuvers

approved

### 2.19 TYPES OF OPERATIONS

The airplane is approved for the following operations when equipped in accordance with FAR 91 or FAR 135.

- (a) Day V.F.R.
- (b) Night V.F.R.
- (c) Day I.F.R.
- (d) Night I.F.R.
- (e) Non Icing

### 2.21 FUEL LIMITATIONS

(a) Total Capacity

77 U.S. GAL.

(b) Unusable Fuel

5 U.S. GAL.

The unusable fuel for this airplane has been determined as 2.5 gallons in each wing tank in critical flight attitudes.

(c) Usable Fuel

72 U.S. GAL.

- The usable fuel in this airplane has been determined as 36.0 gallons in each wing tank.
- (d) Fuel remaining when the quantity indicators read zero cannot be used safely in flight.

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SECTION 2 LIMITATIONS

### 2.23 NOISE LEVEL

The noise level of this aircraft is 75.5 d B(A).

No determination has been made by the Federal Aviation Administration that the noise levels of this airplane are or should be acceptable or unacceptable for operation at, into, or out of, any airport.

The above statement not withstanding the noise level stated above has been verified by and approved by the Federal Aviation Administration in noise level test flights conducted in accordance with FAR 36, Noise Standards - Aircraft Type and Airworthiness Certification. This aircraft model is in compliance with all FAR 36 noise standards applicable to this type.

### 2.25 PLACARDS

In full view of the piiot:

THIS AIRPLANE MUST BE OPERATED AS A NORMAL CATEGORY AIRPLANE IN COMPLIANCE WITH THE OPERATING LIMITATIONS STATED IN THE FORM OF PLACARDS, MARKINGS AND MANUALS.

THIS AIRCRAFT APPROVED FOR NIGHT I.F.R. NON-ICING FLIGHT WHEN EQUIPPED IN ACCORDANCE WITH FAR 91 OR FAR 135.

In full view of the pilot, the following Takeoff and Landing Check Lists will be installed:

## TAKEOFF CHECK LIST

Fuel on Proper Tank
Electric Fuel Pump - On
Engine Gauges - Checked
Alternate Air - Closed
Seat Backs Erect
Mixture - Set
Propeller - Set

Fasten Belts/Harness
Flaps - Set
Controls - Free
Doors - Latched
Air Conditioner - Off

#### LANDING CHECK LIST

Fuel on Proper Tank
Seat Backs Erect
Fasten Belts/Harness
Electric Fuel Pump - On
Mixture - Rich

Propeller - Set
Gear Down
Flaps - Set (White Arc)
Air Conditioner - Off

The "Air Conditioner Off" item in the above Takeoff and Landing Check Lists is mandatory for air conditioned aircraft only.

On the instrument panel in full view of the pilot:

MANEUVERING SPEED 121 KIAS AT 2750 LBS. (SEE A.F.M.)

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SECTION 2 LIMITATIONS

On the instrument panel in full view of the pilot:

#### DEMONSTRATED CROSSWIND COMPONENT 17 KTS

On the instrument panel in full view of the pilot:

NO ACROBATIC MANEUVERS, INCLUDING SPINS, APPROVED

On the instrument panel in full view of the pilot:

GEAR DOWN

130 KIAS (MAX.)

GEAR UP

109 KIAS (MAX.)

**EXTENDED** 

130 KIAS (MAX.)

Near emergency gear lever:

## **EMERGENCY DOWN**

Near emergency gear lever (aircraft equipped with backup gear extender)

OVERRIDE ENGAGED AUTO-EXT-OFFLOCK PIN ON SIDETO ENGAGE OVERRIDE:
PULL LEVER FULL UP, PUSH LOCK PIN
TO RELEASE OVERRIDE:
PULL LEVER FULL UP & RELEASE

Near gear selector switch:

GEAR UP

109 KIAS MAX.

DOWN 130 KIAS MAX.

Adjacent to upper door latch:

ENGAGE LATCH BEFORE FLIGHT

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On the instrument panel in full view of the pilot:

#### WARNING

TURN OFF STROBE LIGHTS WHEN IN CLOSE PROXIMITY TO GROUND OR DURING FLIGHT THROUGH CLOUD, FOG OR HAZE.

In full view of the pilot, in the area of the air conditioner controls when the air conditioner is installed:

#### WARNING

AIR CONDITIONER MUST BE OFF TO INSURE NORMAL TAKEOFF CLIMB PERFORMANCE.

On inside of baggage compartment door:

BAGGAGE MAXIMUM 200 LBS. SEE WEIGHT AND BALANCE DATA FOR BAGGAGE LOADING BETWEEN 150 LBS. AND 200 LBS.

Adjacent to fuel tank filler caps:

FUEL - 100/130 AVIATION GRADE - MIN. USABLE CAPACITY 36 GAL.

USABLE CAPACITY TO BOTTOM OF FILLER NECK INDICATOR 25 GAL.

Above fuel quantity gauges:

FUEL REMAINING WHEN QUANTITY INDICATOR READS ZERO CANNOT BE USED SAFELY IN FLIGHT.

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SECTION 2 LIMITATIONS

On the instrument panel in full view of the pilot in aircraft with McCauley propeller installations only:

AVOID CONTINUOUS OPERATION BETWEEN 1500 AND 1950 RPM BELOW 15" MANIFOLD PRESSURE.

On the aft baggage closeout:

MAXIMUM BAGGAGE 200 LBS. NO HEAVY OBJECTS ON HAT SHELF.

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