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# SECTION 1

# GENERAL

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#### SECTION I

#### GENERAL

#### 1.1 INTRODUCTION

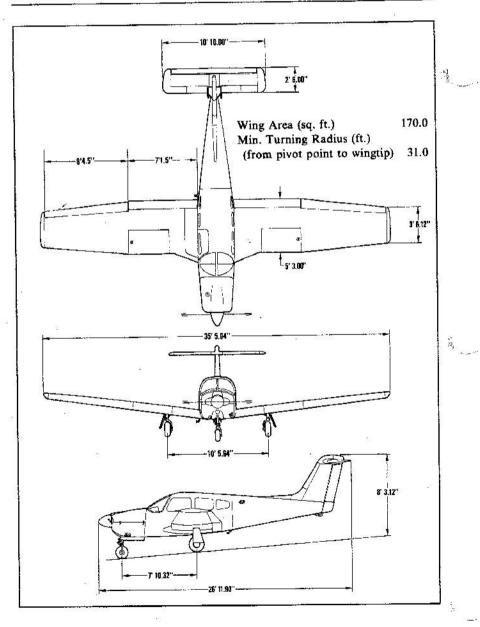
This Pilot's Operating Handbook is designed for maximum utilization as an operating guide for the pilot. It includes the material required to be furnished to the pilot by CAR 3 and FAR Part 21 Subpart J. It also contains supplemental data supplied by the airplane manufacturer.

This handbook is not designed as a substitute for adequate and competent flight instruction, knowledge of current airworthiness directives, applicable federal air regulations or advisory circulars. It is not intended to be a guide for basic flight instruction or a training manual and should not be used for operational purposes unless kept in a current status.

Assurance that the airplane is in an airworthy condition is the responsibility of the owner. The pilot in command is responsible for determining that the airplane is safe for flight. The pilot is also responsible for remaining within the operating limitations as outlined by instrument markings, placards, and this handbook.

Although the arrangement of this handbook is intended to increase its in-flight capabilities, it should not be used solely as an occasional operating reference. The pilot should study the entire handbook to familiarize himself with the limitations, performance, procedures and operational handling characteristics of the airplane before flight.

The handbook has been divided into numbered (arabic) sections, each provided with a "finger-tip" tab divider for quick reference. The limitations and emergency procedures have been placed ahead of the normal procedures, performance and other sections to provide easier access to information that may be required in flight. The "Emergency Procedures" Section has been furnished with a red tab divider to present an instant reference to the section. Provisions for expansion of the handbook have been made by the deliberate omission of certain paragraph numbers, figure numbers, item numbers and pages noted as being intentionally left blank.



THREE VIEW
Figure 1-1

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#### APPLICABILITY

The aircraft serial number eligibility bracket for application of this handbook is 28R-7918001 through 28R-7918267. The specific application of this handbook is limited to the Piper PA-28RT-201 model airplanc designated by serial number and registration number on the face of the title page of this handbook.

This handbook cannot be used for operational purposes unless kept in a current status.

#### REVISIONS

The information compiled in the Pilot's Operating Handbook will be kept current by revisions distributed to the airplane owners.

Revision material will consist of information necessary to update the text of the present handbook and/or to add information to cover added airplane equipment.

#### 1. Revisions

Revisions will be distributed whenever necessary as complete page replacements or additions and shall be inserted into the handbook in accordance with the instructions given below.

- 1. Revision pages will replace only pages with the same page number.
- 2. Insert all additional pages in proper numerical order within each section.
- Page numbers followed by a small letter shall be inserted in direct sequence with the same common numbered page.

## II. Identification of Revised Material

Revised text and illustrations shall be indicated by a black vertical line along the outside margin of the page, opposite revised, added or deleted material. A line along the outside margin of the page opposite the page number will indicate that an entire page was added.

Black lines will indicate only current revisions with changes and additions to or deletions of existing text and illustrations. Changes in capitalization, spelling, punctuation or the physical location of material on a page will not be identified by symbols.

## ORIGINAL PAGES ISSUED

The original pages issued for this handbook prior to revision are given below:

Title, ii through vii, 1-1 through 1-21, 2-1 through 2-11, 3-1 through 3-17, 4-1 through 4-26, 5-1 through 5-31, 6-1 through 6-47, 7-1 through 7-31, 8-1 through 8-17, 9-1 through 9-8, 10-1 through 10-2.

# PIPER AIRCRAFT CORPORATION PA-28RT-201, ARROW IV

SECTION 1 GENERAL

#### 1.3 ENGINES

(a)	Number of Engines	ĩ
(b)	Engine Manufacturer	Lycoming
(c)	Engine Model Number	10-360-C1C6
(d)	Rated Horsepower	200
(e)	Rated Speed (rpm)	2700
<b>(f)</b>	Bore (in.)	5.125
(g)	Stroke (in.)	4.375
(h)	Displacement (cu. in.)	361
(i)	Compression Ratio	8.5:1
<b>(j)</b>	Engine Type	Four Cylinder, Direct
		Drive, Horizontally
	*	Opposed, Air Cooled
		and Fuel Injected

#### 1.5 PROPELLERS

# McCAULEY (a) Number of Propellers (b) Propeller Manufacturer (c) Blade Model (d) Number of Blades (e) Hub Model (f) Propeller Diameter (in.) (l) Maximum (2) Minimum (2) Minimum (3) Propeller Type Constant Speed, Hydraulically Actuated

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## SECTION 1 GENERAL

# PIPER AIRCRAFT CORPORATION PA-28RT-201, ARROW IV

HARTZELL  (a) Number of Propellers  (b) Propeller Manufacturer  (c) Blade Model  (d) Number of Blades  (e) Hub Model  (f) Propeller Diameter (in.)  (l) Maximum  (2) Minimum  (g) Propeller Type	Hartzell F7666A-2R 2 HC-C2YK-1( )F 74 72 Constant Speed, Hydraulically Actuated
1.7 FUEL  (a) Fuel Capacity (U.S. gal.) (total) (b) Usable Fuel (U.S. gal.) (total) (c) Fuel Grade, Aviation (l) Minimum Octane (2) Specified Octane  (3) Alternate Fuels	77 72  100/130 - Green 100 - Green, 100 LL - Blue or 100/130 - Green Refer to latest revision of Lycoming Service Instruction 1070
<ul><li>(a) Oil Capacity (U.S. qts.)</li><li>(b) Oil Specification</li><li>(c) Oil Viscosity</li></ul>	Refer to latest issue of Lycoming Service Instruction 1014 Refer to Section 8 - paragraph 8.19

2750
2750
200
ă.
1627
1123
24
22
20
16.18
13.75

<sup>\*</sup>These values are approximate and vary from one aircraft to another. Refer to Figure 6-5 for the Standard Empty Weight value and the Useful Load value to be used for C.G. calculations for the aircraft specified.

# 1.19 SYMBOLS, ABBREVIATIONS AND TERMINOLOGY

The following definitions are of symbols, abbreviations and terminology used throughout the handbook and those which may be of added operational significance to the pilot.

# (a) General Airspeed Terminology and Symbols

CAS	Calibrated Airspeed means the indicated speed of an aircraft, corrected for position and instrument error. Calibrated airspeed is equal to true airspeed in standard atmosphere at sea level.
KCAS	Calibrated Airspeed expressed in "Knots."
GS	Ground Speed is the speed of an airplane relative to the ground.
IAS	Indicated Airspeed is the speed of an air- craft as shown on the airspeed indicator when corrected for instrument error. IAS values published in this handbook assume zero instrument error.
KIAS	Indicated Airspeed expressed in "Knots."
M	Mach Number is the ratio of true airspeed to the speed of sound.
TAS	True Airspeed is the airspeed of an airplane relative to undisturbed air which is the CAS corrected for altitude, temperature and compressibility.
VA	Maneuvering Speed is the maximum speed at which application of full available aerodynamic control will not overstress the airplane.
VFE	Maximum Flap Extended Speed is the highest speed permissible with wing flaps in a prescribed extended position.

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VLE Maximum Landing Gear Extended Speed is the maximum speed at which an aircraft can be safely flown with the landing gear extended. VLO Maximum Landing Gear Operating Speed is the maximum speed at which the landing gear can be safely extended or retracted. VNE/MNE Never Exceed Speed or Mach Number is the speed limit that may not be exceeded at any time. **VNO** Maximum Structural Cruising Speed is the speed that should not be exceeded except in smooth air and then only with caution. VS Stalling Speed or the minimum steady flight speed at which the airplane is controllable. VSO Stalling Speed or the minimum steady flight speed at which the airplane is controllable in the landing configuration. VXBest Angle-of-Climb Speed is the airspeed which delivers the greatest gain of altitude in the shortest possible horizontal distance. VY Best Rate-of-Climb Speed is the airspeed which delivers the greatest gain in altitude in the shortest possible time.

#### PIPER AIRCRAFT CORPORATION PA-28RT-201, ARROW IV

## (b) Meteorological Terminology

ISA

**SECTION 1** 

GENERAL

International Standard Atmosphere in which: The air is a dry perfect gas; The temperature at sea level is 15° Celsius (59° Fahrenheit); The pressure at sea level is 29.92 inches Hg (1013 mb); The temperature gradient from sea level to the altitude at which the temperature is -56.5° C (-69.7° F) is -0.00198° C (-0.003566° F) per foot and zero above that altitude.

OAT

Outside Air Temperature is the free air static temperature, obtained either from inflight temperature indications or ground meteorological sources, adjusted for instrument error and compressibility effects.

Indicated Pressure Altitude The number actually read from an altimeter when the barometric subscale has been set to 29.92 inches of mercury (1013 millibars).

Pressure Altitude

Altitude measured from standard sea-level pressure (29.92 in. Hg) by a pressure or barometric altimeter. It is the indicated pressure altitude corrected for position and instrument error. In this handbook, altimeter instrument errors are assumed to be zero.

14134040 - 204360

Station Pressure

Actual atmospheric pressure at field elevation.

Wind

The wind velocities recorded as variables on the charts of this handbook are to be understood as the headwind or tailwind components of the reported winds.

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#### (c) Power Terminology

Takeoff Power

Maximum power permissible for takeoff.

Maximum Continuous Power

Maximum power permissible continuously

during flight.

Maximum Climb

Power

Maximum power permissible during

climb.

Maximum Cruise

Power

Maximum power permissible during cruise.

#### (d) Engine Instruments

**EGT** Gauge

Exhaust Gas Temperature Gauge

# (e) Airplane Performance and Flight Planning Terminology

Climb Gradient

The demonstrated ratio of the change in height during a portion of a climb, to the horizontal distance traversed in the same

time interval.

Demonstrated Crosswind Velocity

The demonstrated crosswind velocity is the velocity of the crosswind component for which adequate control of the airplane during takeoff and landing was actually demonstrated during certification tests.

Accelerate-Stop Distance

The distance required to accelerate an airplane to a specified speed and, assuming failure of an engine at the instant that speed is attained, to bring the airplane to a stop.

MEA

Minimum en route IFR altitude.

Route Segment

A part of a route. Each end of that part is identified by: (1) a geographical location; or (2) a point at which a definite radio fix can be established.

#### **SECTION 1 GENERAL**

#### PIPER AIRCRAFT CORPORATION PA-28RT-201, ARROW IV

# (f) Weight and Balance Terminology

Reference Datum

An imaginary vertical plane from which all horizontal distances are measured for

balance purposes.

Station

A location along the airplane fuselage usually given in terms of distance from the

reference datum.

Arm

The horizontal distance from the reference

datum to the center of gravity (C.G.) of an

Moment

The product of the weight of an item multiplied by its arm. (Moment divided by a constant is used to simplify balance calculations by reducing the number of digits.)

Center of Gravity

(C.G.)

The point at which an airplane would balance if suspended. Its distance from the reference datum is found by dividing the total moment by the total weight of the

airplane.

C.G. Arm

The arm obtained by adding the airplane's individual moments and dividing the sum

by the total weight.

C.G. Limits

The extreme center of gravity locations within which the airplane must be operated

at a given weight.

Usable Fuel

Fuel available for flight planning.

Unusable Fuel

Fuel remaining after a runout test has been completed in accordance with govern-

mental regulations.

Standard Empty

Weight

Weight of a standard airplane including unusable fuel, full operating fluids and full

oil.

# PIPER AIRCRAFT CORPORATION PA-28RT-201, ARROW IV

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Basic Empty Weight

Standard empty weight plus optional

equipment.

Payload

Weight of occupants, cargo and baggage.

Useful Load

Difference between takeoff weight, or ramp weight if applicable, and basic empty

weight.

Maximum Ramp

Weight

Maximum weight approved for ground maneuver. (It includes weight of start, taxi

and run up fuel.)

Maximum Takeoff Weight Maximum weight approved for the start of

the takeoff run.

Maximum Landing Weight Maximum weight approved for the landing

touchdown.

Maximum Zero Fuel Weight Maximum weight exclusive of usable fuel.

# 1.21 CONVERSION FACTORS

MULTIPLY	BY	TO OBTAIN
acres	0.4047	ha
	43560	sq. ft.
<i>*</i>	0.0015625	sq. mi.
atmospheres (atm)	76	cm Hg
)	29.92	in. Hg
	1,0133	bar
	1.033	kg/cm <sup>2</sup>
	14.70	lb./sq. in.
	2116	lb./sq. ft.
bars (bar)	0.98692	atm
vars (our)	14.503768	lb./sq. in.
British Thermal Unit (BTU)	0.2519958	kg-cal
centimeters (cm)	0.3937	in.
, , , , , , , , , , , , , , , , , , ,	0.032808	ft.
centimeters of mercury at	0.01316	atm
0°C (cm Hg)	0.3937	in. Hg
0 0 (0 1-8)	0.1934	lb./sq. in.
15	27.85	lb./sq. ft.
4	135.95	$kg/m^{-2}$
centimeters per second	0.032808	ft./sec.
(cm/sec.)	1.9685	ft./min.
(0111, 5001)	0.02237	mph
cubic centimeters (cm <sup>3</sup> )	0.03381	fl. oz.
	0.06102	cu, in.
	3.531 x 10 -5	cu. ft.
	0.001	1
	2.642 x 10 -4	U.S. gal.

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MULTIPLY	BY	TO OBTAIN	
cubic feet (cu. ft.)	28317	cm <sup>3</sup>	
	0.028317	m <sup>3</sup>	
	1728	cu. in.	
	0.037037	cu. yd.	
	7.481	U.S. gal.	
	28.32	1	
cubic feet per minute	0.472	1/sec.	
(cu. ft./min.)	0.028317	$m^{3}/min.$	
cubic inches (cu. in.)	16.39	cm <sup>3</sup>	
	1.639 x 10 -5	m <sup>3</sup>	
	5.787 x 10 -4	cu. ft.	
	0.5541	fl. oz.	
	0.01639	1	
	4.329 x 10 -3	U.S. gal.	
	0.01732	U.S. qt.	
cubic meters (m <sup>3</sup> )	61024	cu. in.	
	1.308	cu. yd.	
	35.3147	cu. ft.	
	264.2	U.S. gal.	
cubic meters per minute (m <sup>3</sup> /min.)	35.3147	cu. ft./min.	
cubic yards (cu. yd.)	27	cu, ft.	
	0.7646	m <sup>3</sup>	
	202	U.S. gal.	
degrees (arc)	0.01745	radians	
degrees per second (deg./sec.)	0.01745	radians/sec.	
lrams, fluid (dr. fl.)	0.125	fl. oz.	
Irams, avdp. (dr. avdp.)	0.0625	oz. avdp.	

SECTION	1
CENERAL	

## PIPER AIRCRAFT CORPORATION PA-28RT-201, ARROW IV

MULTIPLY	BY	TO OBTAIN
feet (ft.)	30.48 0.3048 12 0.33333 0.0606061 1.894 x 10 -4 1.645 x 10 -4	cm m in. yd. rod mi. NM
feet per minute (ft./min.)	0.01136 0.01829 0.508 0.00508	mph km/hr. cm/sec. m/sec.
feet per second (ft./sec.)	0.6818 1.097 30.48 0.5921	mph km/hr. cm/sec. kts.
foot-pounds (ftlb.)	0.138255 3.24 x 10 -4	m-kg kg-cal
foot-pounds per minute (ftlb./min.)	3.030 x 10 -5	hp
foot-pounds per second (ftib./sec.)	1.818 x 10 -5	hp
gallons, Imperial (Imperial gal.)	277.4 1,201 4.546	cu. in. U.S. gal. 1
gallons, U.S. dry (U.S. gal. dry)	268.8 1.556 x 10 -1 1.164 4.405	cu. in. cu. ft. U.S. gal. 1

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		3
MULTIPLY	BY	TO OBTAIN
gallons, U.S. liquid	231	cu. in.
(U.S. gal.)	0.1337	cu. ft.
, ,	4.951 x 10 -3	cu. yd.
	3785.4	cm <sup>3</sup>
	3.785 x 10 -3	m <sup>3</sup>
	3.785	1
	0.83268	Imperial gal.
2	128	fl. oz.
gallons per acre (gal./acre)	9.353	1/ha
grams (g)	100.0	kg
	0.3527	oz. avdp.
	$2.205 \times 10^{-3}$	lb.
grams per centimeter	0.1	kg/m
(g/cm)	6.721 x 10 -2	lb./ft.
£2	5.601 x 10 -3	lb./in.
grams per cubic	1000	kg/m <sup>-3</sup>
centimeter (g/cm <sup>3</sup> )	0.03613	lb./cu. in.
	62.43	lb./cu. ft.
hectares (ha)	2.471	acres
	107639	sq. ft,
	10000	m 2
horsepower (hp)	33000	ftlb./min.
	550	ftlb./sec.
	76.04	m-kg/sec.
	1.014	metric hp
horsepower, metric	75	m-kg/sec.
	0.9863	hp
inches (in.)	25.40	mm
	2.540	em
	0.0254	m
	0.08333	ft.
	0.027777	yd.

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# PIPER AIRCRAFT CORPORATION PA-28RT-201, ARROW IV

MULTIPLY	BY	TO OBTAIN
inches of mercury at 0°C (in. Hg)	0.033421 0.4912 70.73 345.3 2.540 25.40	atm lb./sq. in. lb./sq. ft. kg/m <sup>2</sup> cm Hg mm Hg
inch-pounds (inlb.)	0.011521	m-kg
kilograms (kg)	2.204622 35.27 1000	lb. oz. avdp. g
kilogram-calories (kg-cal)	3.9683 3087 426.9	BTU ftlb. m-kg
kilograms per cubic meter (kg/m <sup>3</sup> )	0.06243 0.001	lb./cu. ft. g/cm <sup>3</sup>
kilograms per hectare (kg/ha)	0.892	lb./acre
kilograms per square centimeter (kg/cm <sup>2</sup> )	0.9678 28.96 14.22 2048	atm in. Hg lb./sq. in. lb./sq. ft.
kilograms per square meter (kg/m <sup>2</sup> )	2.896 x 10 -3 1.422 x 10 -3 0.2048	in. Hg lb./sq. in. lb./sq. ft.
kilometers (km)	1 x 10 -5 3280.8 0.6214 0.53996	cm ft. mi. NM

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MULTIPLY	ВУ	TO OBTAIN
kilometers per hour	0.9113	ft./sec.
(km/hr.)	58.68	ft./min.
	0.53996	kt
	0.6214	mph
	0.27778	m/sec.
	16.67	m/min.
knots (kt)	1	nautical mph
	1.689	ft./sec.
	1.1516	statute mph
	1.852	km/hr.
	51.48	m/sec.
iters (1)	1000	cm <sup>3</sup>
	61.02	cu. in.
	0.03531	cu. ft.
	33.814	fl. oz.
	0.264172	U.S. gal,
	0.2200	Imperial gal.
	1.05669	qt.
iters per hectare	13.69	fl. oz./acre
(1/ha)	0.107	gal./acre
iters per second (1/sec.)	2.12	cu. ft./min.
neters (m)	39.37	in.
100 CO	3.280840	ft.
	1.0936	yd.
	0.198838	rod
	6.214 x 10 -4	mi.
	5.3996 x 10 -4	NM
neter-kilogram	7.23301	ftlb.
(m-kg)	86.798	inlb.
meters per minute (m/min.)	0.06	km/hr.

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MULTIPLY	BY	TO OBTAIN
meters per second (m/sec.)	3.280840 196.8504 2.237 3.6	ft./sec. ft./min. mph km/hr.
microns	3.937 x 10 -5	in.
miles, statute (mi.)	5280 1.6093 1609.3 0.8684	ft. km m NM
miles per hour (mph)	44.7041 4.470 x 10 -1 1.467 88 1.6093 0.8684	cm/sec. m/sec. ft./sec. ft./min. km/hr. kt
miles per hour square (m/hr. sq.)	2.151	ft./sec. sq.
millibars	2.953 x 10 -2	in. Hg
millimeters (mm)	0.03937	in.
millimeters of mercury at 0°C (mm Hg)	0.03937	in. Hg
nautical miles (NM)	6080 1.1516 1852 1.852	ft. statute mi. m km
ounces, avdp. (oz. avdp.)	28.35 16	g dr. avdp.

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PIPER AIRCRAFT CORPO PA-28RT-201, ARROW IV	DRATION	SECTION 1 GENERAL
MULTIPLY	ВУ	TO OBTAIN
ounces, fluid (fl. oz.)	8	dr. fl.
ii 8	29.57	cm <sup>3</sup>
	1.805	cu. in.
	0.0296	1
	0.0078	U.S. gal.
ounces, fluid per acre (fl. oz./acre)	0.073	1/ha
pounds (lb.)	0.453592	kg
	453.6	g
	3.108 x 10 -2	slug
pounds per acre (lb./acre)	1.121	kg/ha
pounds per cubic foot (lb./cu. ft.)	16.02	kg/m <sup>3</sup>
pounds per cubic inch	1728	lb./cu. ft.
(lb./cu. in.)	27.68	g/cm <sup>3</sup>
pounds per square foot	0.1414	in. Hg
(lb./sq. ft.)	4.88243	$kg/m^{-2}$
10	4.725 x 10 -4	atm
pounds per square inch	5.1715	cm Hg
(psi or lb./sq. in.)	2.036	in. Hg
	0.06804	atm
8	0.0689476	bar
	703.1	kg/m <sup>2</sup>
quart, U.S. (qt.)	0.94635	1
	57.749	cu. in.
radians	57.30	deg. (arc)
	0.1592	rev.
radians per second	57.30	deg./sec.
(radians/sec.)	0.1592	rev./sec.
	9.549	rpm

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MULTIPLY	BY	TO OBTAIN
revolutions (rev.)	6.283	radians
revolutions per minute (rpm or rev./min.)	0.1047	radians/sec.
revolutions per second (rev./sec.)	6.283	radians/sec.
rod	16.5 5.5 5.029	ft. yd. m
slug	32.174	lb.
square centimeters (cm <sup>2</sup> )	0.1550 0.001076	sq. in. sq. ft.
square feet (sq. ft.)	929 0.092903 144 0.1111 2.296 x 10 -5	cm <sup>2</sup> m <sup>2</sup> sq. in. sq. yd. acres
square inches (sq. in.)	6.4516 6.944 x 10 - <sup>3</sup>	cm <sup>2</sup> sq. ft.
square kilometers (km <sup>2</sup> )	0.3861	sq. mi.
square meters (m <sup>2</sup> )	10.76391 1.196 0.0001	sq. ft. sq. yd. ha
square miles (sq. mi.)	2.590 640	km <sup>2</sup> acres
square rods (sq. rods)	30.25	sq. yd.
square yards (sq. yd.)	0.8361 9 0.0330579	m <sup>2</sup> sq. ft. sq. rods

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MULTIPLY	ВУ	TO OBTAIN
yards (yd.)	0.9144	m
	36	ft. in.
	0.181818	rod

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# SECTION 2

# LIMITATIONS

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