# TABLE OF CONTENTS

# **SECTION 2**

# LIMITATIONS

Paragra	ph	Page
No.		_
		No.
2.1	General	2.1
2.3	Airspeed Limitations	2-1
2.5	Airspeed Indicator Markings	2-1
2.7	Power Plant Limitations	2-2
2.9	Power Plant Instrument Markings	2-3 2-4
2.11	Weight Limits	2-4
2.13	Center of Gravity Limits	2-4
2.15	Maneuver Limits	2-6
2.17	Flight Load Factors	2-6
2.19	Types of Operation	2-6
2.21	Fuel Limitations	2-6
2.23	Noise Level	2-0 2-7
2.25	Placards	2-7

REPORT: VB-1120

# TABLE OF CONTENTS SECTION 2 LIMITATIONS

Paragraph No. Page No.

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REPORT: VB-1120

2-ii

#### **SECTION 2**

#### LIMITATIONS

# 2.1 GENERAL

This section provides the ''FAA Approved'' operating limitations, instrument markings, color coding and basic placards necessary for operation of the airplane and its systems.

This airplane must be operated as a normal or utility category airplane in compliance with the operating limitations stated in the form of placards and markings and those given in this section and this complete handbook.

Limitations associated with those optional systems and equipment which require handbook supplements can be found in Section 9 (Supplements).

# 2.3 AIRSPEED LIMITATIONS

SPEED	KIAS	KCAS
Never Exceed Speed (VNE) - Do not exceed this speed in any operation.	154	148
Maximum Structural Cruising Speed (VNO) - Do not exceed this speed except in		
smooth air and then only with caution.	125	121

ISSUED: JULY 2, 1979 REVISED: JULY 21, 1982

REPORT: VB-1120

#### SECTION 2 LIMITATIONS

# PIPER AIRCRAFT CORPORATION PA-28-181, ARCHER II

SPEED	KIAS	KCAS
Design Maneuvering Speed (VA) - Do not make full or abrupt control movements above this speed.		ž
At 2550 lbs. G.W.	113	111
At 1634 lbs. G.W.	89	89
CAUTION		

Maneuvering speed decreases at lighter weight as the effects of aerodynamic forces become more pronounced. Linear interpolation may be used for intermediate gross weights. Maneuvering speed should not be exceeded while operating in rough air.

Maximum Flaps Extended Speed (VFE) - Do not exceed this speed with the flaps extended.

102

100

#### 2.5 AIRSPEED INDICATOR MARKINGS

MARKING	IAS
Red Radial Line (Never Exceed)	154 KTS
Yellow Arc (Caution Range - Smooth Air Only)	125 KTS to 154 KTS
Green Arc (Normal Operating Range)	55 KTS to 125 KTS
White Arc (Flap Down)	49 KTS to 102 KTS

REPORT: VB-1120

**ISSUED: JULY 2, 1979** 

# PIPER AIRCRAFT CORPORATION PA-28-181, ARCHER II

# SECTION 2 LIMITATIONS

# 2.7 POWER PLANT LIMITATIONS

TOWER TEAMT LIMITATIONS	
(a) Number of Engines	1
(b) Engine Manufacturer	I voomin -
(c) Engine Model No.	Lycoming O-360-A4M or
as at G-4 D-boson's bloods	O-360-A4M or
	carburetor setting
(d) Engine Operating Limits	10-3878
(1) Takeoff Power - 5 Minute	
limit (BHP)	190
(2) Takeoff Engine Speed - 5	180
Minute Limit (RPM)	2700
(3) Maximum Continuous Power	2700
(BHP)	178
(4) Maximum Continuous Engine	178
Speed (RPM)	2650
(5) Maximum Oil Temperature	245°F
(6) Oil Pressure	243 1
Minimum (red line)	25 PSI
Maximum (red line)	90 or 100 PSI
(7) Fuel Pressure	30 01 100 1 BI
Minimum (red line)	0.5 PSI
Maximum (red line)	8 PSI
(8) Fuel (AVGAS ONLY)	
(minimum grade)	100 or 100LL
	Aviation Grade
(9) Number of Propellers	1
(10) Propeller Manufacturer	Sensenich
(11)Propeller Model	76EM8S5-0-62
(12) Propeller Diameter	
Minimum	76 IN.
Maximum	76 IN.
(13) Propeller Tolerance (static RPM	
at maximum permissible throttle	
setting, sea level, ISA)	Not above 2340 RPM
	Not below 2240 RPM

#### NOTE

Refer to the airplane maintenance manual for test procedure to determine approved static rpm under non-standard conditions.

ISSUED: JULY 2, 1979 REVISED: APRIL 2, 1998

REPORT: VB-1120

#### 2.9 POWER PLANT INSTRUMENT MARKINGS

(a)	Tachometer	
	Green Arc (Normal Operating Range)	500 to 2650 RPM
	Yellow Arc (5 Minute Limit)	2650 to 2700 RPM
	Red Line (Takeoff Power)	2700 RPM
(b)	Oil Temperature	
	Green Arc (Normal Operating Range)	75° to 245°F
	Red Line (Maximum)	245°F
(c)	Oil Pressure	
0.2	Green Arc (Normal Operating Range)	60 PSI to 90 PSI
	Yellow Arc (Caution Range) (Idle)	25 PSI to 60 PSI
	Yellow Arc (Ground Warm-Up)	None or 90 PSI to 100 PSI
	Red Line (Minimum)	25 PSI
	Red Line (Maximum)	90 or 100 PSI
(d)	Fuel Pressure	
	Green Arc (Normal Operating Range)	0.5 PSI to 8 PSI
	Red Line (Minimum)	0.5 PSI
	Red Line (Maximum)	8 PSI

#### 2.11 WEIGHT LIMITS

		Normal	Utility
(a)	Maximum Ramp (lbs.)	2558	2138
'	Maximum Weight (lbs.)	2550	2130
. ,	Maximum Baggage (lbs.)	200	0

NOTE

Refer to Section 5 (Performance) for maximum weight as limited by performance.

REPORT: VB-1120

2-4

ISSUED: JULY 2, 1979 REVISED: JULY 21, 1982

# 2.13 CENTER OF GRAVITY LIMITS

#### (a) Normal Category

Weight	Forward Limit	Rearward Limit
Pounds	Inches Aft of Datum	Inches Aft of Datum
2550 2050 (and less)	88.6 82.0	93.0 93.0

# (b) Utility Category

Weight	Forward Limit	Rearward Limit
Pounds	Inches Aft of Datum	Inches Aft of Datum
2130 2050 (and less)	83.0 82.0	93.0 93.0

#### **NOTES**

Straight line variation between points given.

The datum used is 78.4 inches ahead of the wing leading edge at the inboard intersection of the straight and tapered section.

It is the responsibility of the airplane owner and the pilot to insure that the airplane is properly loaded. See Section 6 (Weight and Balance) for proper loading instructions.

**ISSUED: JULY 2, 1979** 

**REPORT: VB-1120** 

#### PIPER AIRCRAFT CORPORATION PA-28-181, ARCHER II

#### 2.15 MANEUVER LIMITS

- (a) Normal Category All acrobatic maneuvers including spins prohibited.
- (b) Utility Category Approved maneuvers for bank angles exceeding  $60^{\circ}$ .

	Entry Speed
Steep Turns	113 KIAS
Lazy Eights	113 KIAS
Chandelles	113 KIAS

#### 2.17 FLIGHT LOAD FACTORS

	Normal	Utility
(a) Positive Load Factor (Maximum)	3.8 G	4.4 G
(b) Negative Load Factor (Maximum)	No inver	ted maneuvers
		approved

#### 2.19 TYPES OF OPERATION

The airplane is approved for the following operations when equipped in accordance with FAR 91 or FAR 135.

- (a) Day V.F.R.
- (b) Night V.F.R.
- (c) Day I.F.R.
- (d) Night I.F.R.
- (e) Non Icing

#### 2.21 FUEL LIMITATIONS

(a)	Total Capacity	50 U.S. GAL.
(b)	Unusable Fuel	2 U.S. GAL.
20.00	The unusable fuel for this airplane has	
	been determined as 1.0 gallon in each	
	wing in critical flight attitudes.	
(c)	Usable Fuel	48 U.S. GAL.
	The usable fuel in this airplane has been	
	determined as 24.0 gallons in each wing.	

REPORT: VB-1120

ISSUED: JULY 2, 1979

# 2.23 NOISE LEVEL

The noise level of this aircraft is 73.9 d B(A).

No determination has been made by the Federal Aviation Administration that the noise levels of this airplane are or should be acceptable or unacceptable for operation at, into, or out of, any airport.

The above statement not withstanding the noise level stated above has been verified by and approved by the Federal Aviation Administration in noise level test flights conducted in accordance with FAR 36, Noise Standards - Aircraft Type and Airworthiness Certification. This aircraft model is in compliance with all FAR 36 noise standards applicable to this type.

**ISSUED: JULY 2, 1979** 

**REPORT: VB-1120** 

#### 2.25 PLACARDS

In full view of the pilot:

"THIS AIRPLANE MUST BE OPERATED AS A NORMAL OR UTILITY CATEGORY AIRPLANE IN COMPLIANCE WITH THE OPERATING LIMITATIONS STATED IN THE FORM OF PLACARDS, MARK- INGS AND MANUALS.

ALL MARKINGS AND PLACARDS ON THIS AIR-PLANE APPLY TO ITS OPERATION AS A UTILITY CATEGORY AIRPLANE. FOR NORMAL AND UTILITY CATEGORY OPERATION REFER TO THE PILOT'S OPERATING HANDBOOK.

NO ACROBATIC MANEUVERS ARE APPROVED FOR NORMAL CATEGORY OPERATIONS. SPINS ARE PROHIBITED FOR NORMAL AND UTILITY CATEGORY."

In full view of the pilot:

#### TAKEOFF CHECK LIST

Fuel on proper tank

Seat backs erect

Electric fuel pump on

Fasten belts/harness

Engine gauges checked

Trim tab - set

Flaps - set

Controls- free

Carb. heat off

Door - latched

Mixture set

Air Conditioner off

Primer locked

# LANDING CHECK LIST

Fuel on proper tank

Flaps - set

Mixture rich

Fasten belts/harness

Electric fuel pump on

Air Conditioner off

Seat backs erect

The ``AIR COND OFF" item in the above takeoff and landing check lists is mandatory for air conditioned aircraft only.

REPORT: VB-1120

ISSUED: JULY 2, 1979

In full view of the pilot, in the area of the air conditioner control panel when the air conditioner is installed:

"WARNING = AIR CONDITIONER MUST BE OFF TO INSURE NORMAL TAKEOFF CLIMB PER-FORMANCE."

Adjacent to upper door latch:

"ENGAGE LATCH BEFORE FLIGHT."

On inside of the baggage compartment door.

"BAGGAGE MAXIMUM 200 LBS."
"UTILITY CATEGORY OPERATION - NO BAGGAGE OR AFT PASSENGERS ALLOWED. NORMAL CATEGORY OPERATION - SEE PILOT'S OPERATING HANDBOOK WEIGHT AND BALANCE SECTION FOR BAGGAGE AND AFT PASSENGER LIMITATIONS."

In full view of the pilot:

"VA 113 KIAS AT 2550# (SEE P.O.H.)"

"DEMO. X-WIND 17 KTS."

In full view of the pilot:

"OIL COOLER WINTERIZATION PLATE TO BE REMOVED WHEN AMBIENT TEMPERATURE EXCEEDS 50°F."

ISSUED: JULY 2, 1979 REVISED: JULY 21, 1982

REPORT: VB-1120

# PIPER AIRCRAFT CORPORATION PA-28-181, ARCHER II

#### In full view of the pilot:

- "UTILITY CATEGORY OPERATION ONLY."
- (1) NO AFT PASSENGERS ALLOWED.
- (2) ACROBATIC MANEUVERS ARE LIMITED TO THE FOLLOWING:

ENTRY SPEE
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SPINS PROHIBITED	
STEEP TURNS	113 KIAS
LAZY EIGHTS	113 KIAS
CHANDELLES	113 KIAS

#### In full view of the pilot:

"WARNING = TURN OFF STROBE LIGHTS WHEN IN CLOSE PROXIMITY TO GROUND OR DURING FLIGHT THROUGH CLOUD, FOG OR HAZE."

#### On tachometer face:

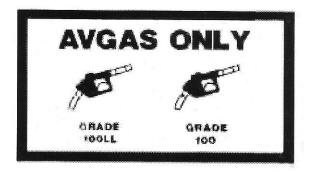
"AFTER 5 MIN: REDUCE POWER TO 2650 RPM."

REPORT: VB-1120 2-10

ISSUED: JULY 2, 1979 REVISED: JULY 20, 1983 Adjacent to the fuel filler caps:

FUEL - 100 or 100LL AVIATION GRADE. or FUEL - 100-130 AVIATION GRADE MIN. USABLE CAPACITY 24 GAL. USABLE CAPACITY TO BOTTOM OF FILLER NECK INDICATOR 17 GAL.

Adjacent to the filler caps (serial numbers 28-8390036 and up):



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REPORT: VB-1120